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Abstract In this study, the problem of a thin film loaded at both ends and in contact with a microstructured substrate modelled by the couple stress theory of elasticity is investigated. To appropriately define the microstructural contact conditions the contribution of couple tractions is considered. By using the Green's functions for a tangential point force and a couple acting on the surface of a couple stress elastic half-plane, the extended boundary conditions, which enforce compatibility between the displacement and rotations along the contact region, provide two singular integral equations. One of them yields an explicit relation

for couple tractions in terms of the interfacial shear stress. By assuming a series of Chebyshev polynomials for the distributions of shear stress along the contact region and using a collocation method, the remaining integral equation is reduced to an algebraic system for the Chebyshev series coefficients. Finally, the axial load in the thin film is computed while varying the characteristic length of the substrate, revealing significant deviations from the classical elastic solution.

Keywords
(separated by '-')

Contact mechanics - Couple stress elasticity - Thin film



Effects of Couple Traction on Contact Problems at the Microscale

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AQ3

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1 Introduction

Microstructured surfaces are crucial in improving the efficiency of electrolytic cells by enhancing reaction kinetics, improving mass transport, and reducing energy losses. Their increased surface area provides more sites for electrochemical reactions to occur, leading to faster reactions and higher current densities within the same geometric footprint. In proton exchange membrane (PEM) fuel cells, these surfaces also enhance the performance of catalyst layers. Additionally, the rise of additive manufacturing has brought significant attention to thin film/substrate systems in micro- and nano-electromechanical systems (MEMS/NEMS). In these applications, the characteristic length scale of the structure may become comparable to that of the material's intrinsic length, making classical elasticity theory insufficient for capturing size-dependent behavior. Atomistic simulations and lattice models can address these limitations but are computationally

expensive. A more practical alternative is to use enhanced constitutive theories that allow for introducing a characteristic length, such as the couple stress elasticity (CSE) theory. According to this theory, the interaction at the microscale between the bodies in contact is described by a richer set of tractions including higher-order stress components, like couple tractions. Recently, this theory has been applied to investigate the size-dependent contact behavior of thin films on substrates. For instance, the contact problem between an Euler-Bernoulli microbeam and a CSE substrate was studied in [1], assuming the interaction occurs via couple tractions. Under this assumption, a strong size effect was observed on the internal shear forces and bending moments of the microbeam when its length is comparable to the characteristic length of the substrate. Subsequently, the study was extended in [2] to consider a receding contact between the beam and the substrate. It was found that incorporating size dependency results in highly peaked contact pressures near the load application point, and it alters the contact pressure distribution significantly when the characteristic lengths of the substrate and contact zone are comparable. Here, we present an application to the problem of a thin film loaded at both ends and bonded to a microstructured substrate subject to a uniform strain resulting from remote mechanical or thermal loading conditions.

2 Governing Equations

The problem of a thin film of length $2a$ and thickness h perfectly bonded to a CSE substrate is shown in Fig. 1. The line loads Q_1 and Q_2 are applied to the right and left edges of the film, respectively. The interaction between the film and the substrate is modeled by the interfacial shear stress, $\tau = \tau_{yx}$, and the couple tractions, $m = m_{yz}$, distributed along the interface, namely for $-a < x < a$, while the peeling stress σ_{yy} is neglected due to the thin film approximation [3]. The Green's functions for a tangential point force and a couple acting upon the top of the CSE half-plane are used for setting the microstructural contact conditions. The tangential displacement u_x at the surface of a couple stress elastic half-plane due to a concentrated tangential load F and a counterclockwise couple C acting at the origin of the coordinate system was obtained in [1, 4], namely

$$u_x(x, 0) = F \frac{1 - \nu_s}{\pi \mu_s} \int_0^{\infty} \frac{\sqrt{1 + s^2 l^2}}{s g(sl)} \cos sx \, ds - C \frac{1 - \nu_s}{\pi \mu_s} \int_0^{\infty} \frac{\sqrt{1 + s^2 l^2} - sl}{g(sl)} \cos sx \, ds, \quad (1)$$

where μ_s , ν_s , and l are the elastic shear modulus, the Poisson's ratio, and the characteristic length of the CSE substrate, respectively, and

$$g(z) = \sqrt{1 + z^2} + 4(1 - \nu_s)z^2(\sqrt{1 + z^2} - z). \quad (2)$$

The corresponding axial strain along the half-plane surface then follows from the derivative of the displacement (1) with respect to the variable x as

$$\varepsilon_{xx}^s(x, 0) = -F \frac{1 - \nu_s}{\pi \mu_s} \int_0^{\infty} \frac{\sqrt{1 + s^2 l^2}}{g(sl)} \sin sx \, ds + C \frac{1 - \nu_s}{\pi \mu_s} \int_0^{\infty} \frac{\sqrt{1 + s^2 l^2} - sl}{g(sl)} s \sin sx \, ds. \quad (3)$$

The balance equations for the thin film require (Fig. 1)

$$\tau(x) = N'(x), \quad m(x) = -t(x)h/2 \quad (4)$$

where $\tau(x)$ and $m(x)$ are the shear tractions and couple tractions exchanged between the thin film and the couple stress substrate, N is the axial load in the thin film and the prime denotes the derivative with respect to x . Let us now consider a thin film of length $2a$ and thickness $h \ll a$ perfectly bonded to the couple stress substrate (Fig. 1), under plane strain conditions. The axial load in the thin film and the corresponding axial strain follow from the integration of eqn (4)₁ between $-a$ and x , using the boundary conditions $N(-a) = Q_2$:

$$N(x) = Q_2 + \int_{-a}^x \tau(t) dt, \quad \varepsilon_{xx}^f(x) = \frac{1 - \nu_f^2}{E_f h} N(x) = \frac{1 - \nu_f^2}{E_f h} \left[Q_2 + \int_{-a}^x \tau(t) dt \right], \quad (5)$$

where E_f and ν_f are the elastic Young modulus and the Poisson's ratio of the elastic thin film, respectively. For the whole film one has $N(a) = Q_1$. By using the Green's function (3), the surface strain on the top of the substrate loaded by the shear and couple tractions, $\tau(x)$ and $m(x)$ as sketched in Fig. 1, is given by

$$\varepsilon_{xx}^s(x, 0) = -\frac{1 - \nu_s}{\pi \mu_s} \left\{ \int_{-a}^a \tau(t) dt \int_0^\infty \frac{\sqrt{1 + s^2 l^2}}{g(s l)} \sin[s(x - t)] ds - \int_{-a}^a m(t) dt \int_0^\infty \frac{\sqrt{1 + s^2 l^2} - s l}{g(s l)} s \sin[s(x - t)] ds \right\} + \varepsilon_{xx}^0, \quad (6)$$

where ε_{xx}^0 is the thermal strain of the substrate. The strain compatibility condition between the film and the substrate then requires: $\varepsilon_{xx}^s(x, 0) = \varepsilon_{xx}^f(x)$, namely, by using Eqs. (5)₂, (6), and the balance condition (4)₂:

$$\frac{\lambda}{2h} \left[Q_2 + \int_{-a}^x \tau(t) dt \right] = -\frac{1}{2\pi} \int_{-a}^a \tau(t) dt \int_0^\infty \frac{2\sqrt{1 + s^2 l^2} + (\sqrt{1 + s^2 l^2} - s l) sh}{g(s l)} \sin[s(x - t)] ds + \frac{\mu_s \varepsilon_{xx}^0}{1 - \nu_s}, \quad (7)$$

where $\lambda = \mu_s(1 - \nu_f)/[\mu_s(1 - \nu_s)]$ is the relative stiffness ratio. By performing an asymptotic expansion of the kernel function as $s \rightarrow \infty$ to single out the most singular terms, as in [3], and introducing the non-dimensional quantities: $H = h/a$, $\Lambda = la$, $z = sa$, $\xi = x/a$, $\eta = t/a$, and $\rho = Q_2/Q_1$, then, the integral equation (7) becomes

$$\frac{\lambda}{2H} \int_{-1}^{\xi} \tau(\eta) d\eta + \int_{-1}^1 \kappa(\xi - \eta) \tau(\eta) d\eta + \frac{1}{3 - 2\nu_s} \left\{ \frac{1}{\pi} \int_{-1}^1 \frac{\tau(\eta)}{\xi - \eta} d\eta + \frac{H}{8\Lambda^2} \left[\int_{-1}^{\xi} \tau(\eta) d\eta - \int_{\xi}^1 \tau(\eta) d\eta \right] \right\} = \frac{\mu_s \varepsilon_{xx}^0}{1 - \nu_s} - \frac{\lambda \rho Q_1}{2Ha}, \quad (8)$$

where

$$\kappa(\xi) = \frac{1}{2\pi(3 - 2\nu_s)} \int_0^\infty \left[4(1 - \nu_s) - \frac{H}{2\Lambda^2 z^2} \right] \frac{(1 - 2\Lambda^2 z^2) \sqrt{1 + \Lambda^2 z^2} + 2\Lambda^3 z^3}{g(z\Lambda)} \sin z\xi dz. \quad (9)$$

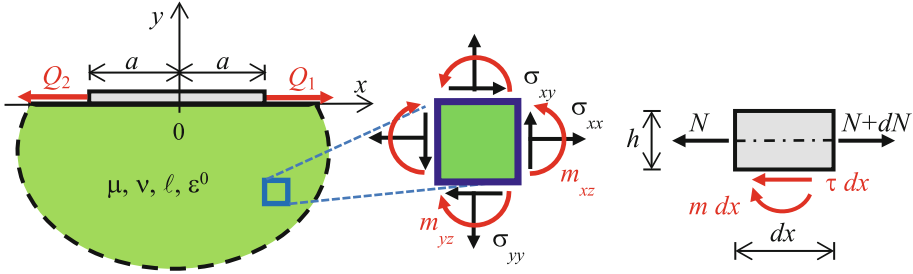


Fig. 1. Thin film attached to a CSE substrate loaded by line loads Q_1 and Q_2 , and free body diagram of forces and couples acting on any infinitesimal thin film element of thickness h .

2.1 Collocation Procedure

At the edges of the contact $x = \pm a$, the shear stress τ must display square root singularity. According to Eq. (4)₂, the couple tractions m also displays square root singularity at $\xi = \pm a$. Therefore, we represent the shear stress by using a truncated series of Chebyshev polynomials of the first kind T_n , as in [3–6] and [7]:

$$\tau(\eta) = \frac{Q_1}{a} \sum_{n=0}^K c_n \frac{T_n(\eta)}{\sqrt{1-\eta^2}}, \quad \text{for } |\eta| \leq 1. \quad (10)$$

Then, from the balance condition for the thin film, $N(a) = Q_1$, one obtains $c_0 = (1-\rho)/\pi$. The unknown coefficients c_n ($n = 1, 2, \dots, K$) can be calculated by using a collocation procedure, namely by solving a linear system of K algebraic equations obtained from the governing integral Eq. (8) evaluated at K collocation points ξ_k ($k = 1, 2, \dots, K$), which are taken as the roots of the Chebyshev polynomial $T_{2K}(\xi)$, namely

$$\sum_{n=1}^K c_n \left\{ \frac{2\pi}{M} \sum_{j=1}^M \kappa(\xi_k - \eta_j) T_n(\eta_j) - \left[\frac{2}{3-2\nu_s} + \left(\frac{H}{(6-4\nu_s)\Lambda^2} + \frac{\lambda}{H} \right) \frac{\sqrt{1-\xi_k^2}}{n} \right] U_{n-1}(\xi_k) \right\} + (1-\rho) \left[\frac{2}{M} \sum_{j=1}^M \kappa(\xi_k - \eta_j) + \left(\frac{H}{(6-4\nu_s)\Lambda^2} + \frac{\lambda}{H} \right) \left(\frac{1}{2} - \frac{\arccos \xi_k}{\pi} \right) \right] = 2\gamma - \frac{\lambda(1+\rho)}{2H}, \quad (11)$$

where

$$\gamma = \frac{a \mu_s \varepsilon_{xx}^0}{(1-\nu_s) Q_1}, \quad \xi_k = \cos \frac{(2k-1)\pi}{2K}, \quad \eta_j = \cos \frac{(2j-1)\pi}{2M}, \quad (12)$$

for $k = 1, 2, \dots, K$, and $j = 1, 2, \dots, M$. Note that the definite integral in Eq. (8) containing the kernel function $\kappa(\xi - \eta)$ has been calculated in Eq. (11) by using the Gauss-Chebyshev quadrature with M nodes η_j ($j = 1, 2, \dots, M$) coinciding with the roots of the Chebyshev polynomial $T_M(\eta)$ defined in (12)₃. Once all the coefficients c_n , for $n = 0, 2, \dots, K$, are calculated, the interfacial shear stress is given by (10) and the axial load in the thin film follows from Eqs. (5) and (10) as

$$N(\xi) = Q_2 + Q_1 \sum_{n=0}^K c_n \int_{-1}^{\xi} \frac{T_n(\eta)}{\sqrt{1-\eta^2}} d\eta = \left[\rho + (1-\rho) \left(1 - \frac{\arccos \xi}{\pi} \right) - \sum_{n=1}^{\infty} \frac{c_n}{2n} U_{n-1}(\xi) \sqrt{1-\xi^2} \right] Q_1. \quad (13)$$

The strength of the shear stress singularity in the CSE substrate at the right and left film edges, k_1 and k_2 , are defined by

$$k_1 = \lim_{x \rightarrow a} \sqrt{2(a-x)} \tau(x), \quad k_2 = \lim_{x \rightarrow -a} \sqrt{2(a+x)} \tau(x), \quad (14)$$

respectively. By introducing the shear stress representation (10), they become:

$$k_1 = \frac{Q_1}{a} \sum_{n=0}^K c_n, \quad k_2 = \frac{Q_1}{a} \sum_{n=0}^K (-1)^n c_n. \quad (15)$$

3 Results

In the following, a Poisson's ratio $\nu_s = 0.3$ is assumed both for the film and substrate, and a number $N \geq 50$ of terms is taken in the Chebyshev series expansion (10). Two special loading case are considered, namely symmetric loading for $\rho = 1$ ($Q_1 = Q_2 = Q$). And skew-symmetric loading for $\rho = -1$ ($Q_1 = -Q_2 = Q$) and $\gamma = 0$. The case of $\rho = 0$ and $\gamma = 0$ can be found in [3]. Note that the shear stress distribution becomes skew-symmetric under symmetric loading and thus only odd terms are considered in the series expansion (10), vice versa the shear stress distribution is symmetric under skew-symmetric loading, so that only even terms are considered.

The effects of the size parameter, l/a , both on the normalized shear stress distribution along the contact zone, $\tau a/Q$, and the axial load in the thin film, N/Q , are illustrated in Fig. 2 under two equal symmetric line loads Q applied at the film edges, for $a/h = 30$ and $\lambda = 1/3$. The shear stress distribution for classical elastic behavior of the substrate obtained from the approach presented in [5] is also reported in Fig. 2 (black dashed line). As expected, the shear stress distribution attains the largest value near the loaded edges of the film, where it displays square root singularity. Introducing the size effect, namely increasing the ratio l/a , the axial load is transferred to the CSE substrate at a smaller distance to the loaded edges. No significant changes are observed in the shear stress and axial load in the film for $l/a > 0.5$. Moreover, the results approach the classical elastic solution [5] for a small but finite value of the characteristic length, thus validating the present approach.

The effects of the substrate strain due to mechanical or thermal loading are considered in Fig. 3 for $\gamma = 1$, and for different values of the characteristic length ratio l/a . In this case, the shear stress distribution is similar to that obtained for $\gamma = 0$ in Fig. 2, and the axial load in the thin film is a bit larger due to the contribution of the substrate strain.

The normalized shear stress distribution, $\tau a/Q$, and axial load, N/Q , under two skew-symmetric line loads Q applied at the film edges, are plotted in Fig. 4, for $a/h = 30$, $\lambda = 1/15$ and for various characteristic length ratios l/a . The classic elastic solution obtained from [5] is also plotted (black dashed lines). The shear stress distribution is symmetric and displays square root singularity near the loaded edges of the film. As the ratio l/a increases, the shear stress decreases in the central part of the film and the axial load is transferred to the CSE substrate at a smaller distance to the loaded edges. No significant changes are observed for $l/a > 0.5$. The classical elastic solution is recovered for a small

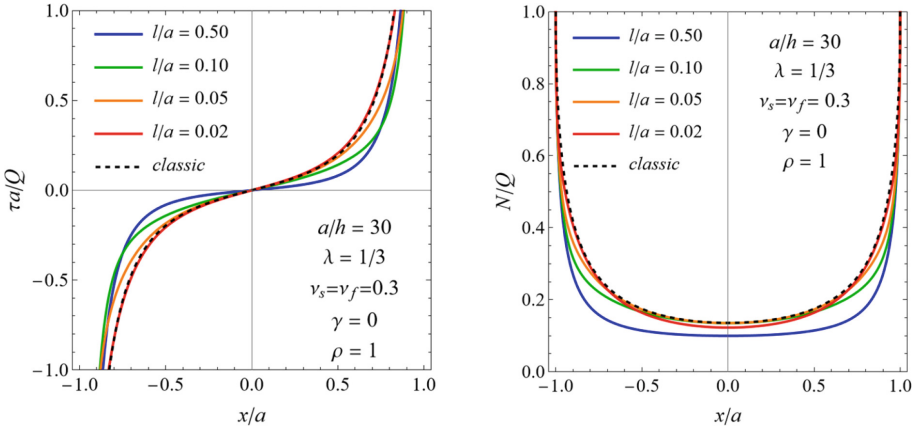


Fig. 2. Normalized distributions of the interfacial shear stress and axial load in the thin film for various values of the ratio l/a under symmetric loading ($Q_1 = Q_2 = Q$).

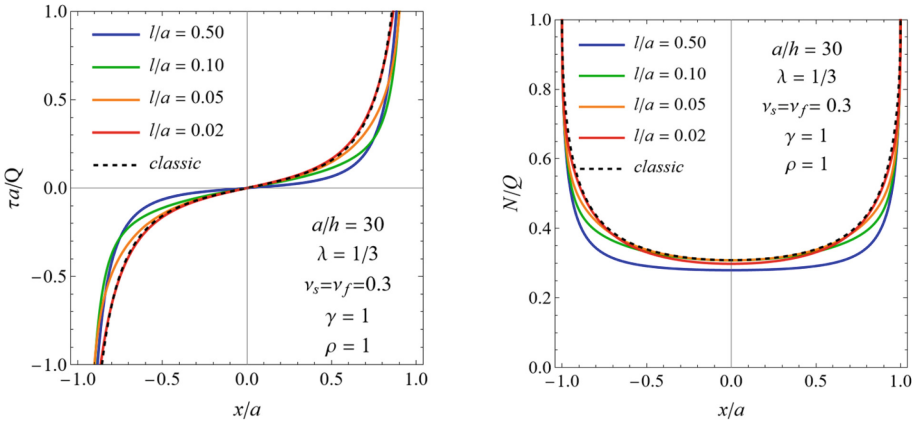


Fig. 3. Normalized distributions of the interfacial shear stress and axial load in the thin film for various values of the ratio l/a under symmetric loading and substrate strain.

but finite value of the ratio l/a . Indeed, for $l = 0$ the couple tractions also vanish along the interface, then for a finite film thickness $h > 0$ the balance condition (4)₂ cannot be satisfied for $\tau(x) \neq 0$ if the bending stiffness of the film is neglected.

Finally, the variations of the strength of stress singularity are presented in Fig. 5 for $a/h = 30$ and $\lambda = 1/15$, by varying the characteristic length, both for symmetric and skew-symmetric loading conditions. The strength of stress singularity is almost unaffected by the characteristic length of the material under symmetric loading, whereas it significantly increases with the characteristic length under skew-symmetric loading, and it attains an almost constant value for $l/a > 0.2$.

According to [5], the normalized strength of stress singularity $k_1 a/Q$ for classical elastic behavior of the substrate with $a/h = 30$ and $\lambda = 1/15$, is equal to 0.648 under

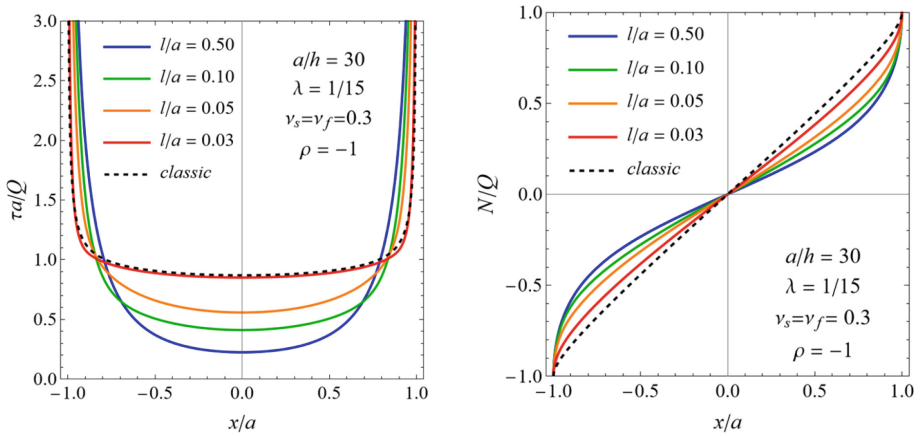


Fig. 4. Normalized distributions of the interfacial shear stress and axial load in the thin film for various values of the ratio l/a under skew symmetric loading ($Q_1 = -Q_2 = Q$).

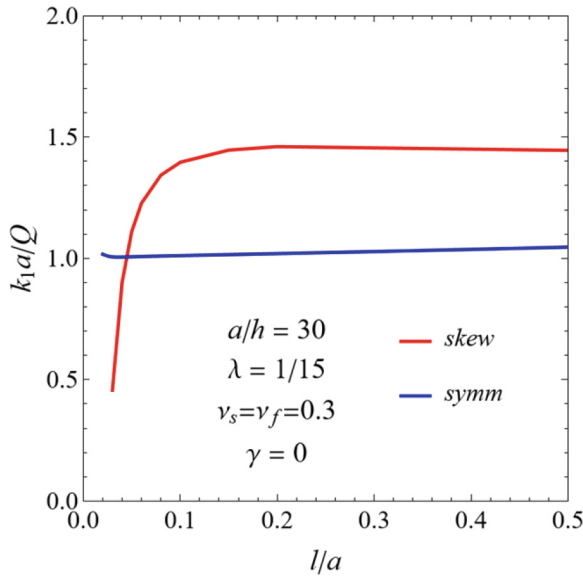


Fig. 5. Normalized variations of the strength of stress singularities k_1 with the characteristic length ratio l/a under symmetric and skew symmetric loading.

symmetric loading and to 0.291 under skew-symmetric loading. Therefore, accounting for the substrate microstructures yields a significant increase in the strength of stress singularity at the film edges.

4 Conclusions

This work presents an analytical model based on Chebyshev series expansions to investigate the contact problem between a thin elastic film under tension and a CSE substrate. The most important features of the present study are:

- the capability to calculate shear and couple tractions along the interface, the axial stress at any point in the thin film, and the strength of stress singularity at the film ends under general loading;
- the use of Chebyshev series expansions for solving a singular integral equation instead of purely numerical procedures;
- the parametric nature of the analysis and the possibility of using it for any geometry and constitutive parameters of the film and the substrate;
- the capability of the model to account for the film thickness in the balance equation of the film.

The present work demonstrates that accounting for non-classical and size-dependent behavior of the substrate as well as for couple tractions yields a remarkable reduction of the axial load and axial stress within the thin film. Moreover, the results provided by the present model prove that the couple tractions significantly contribute to the contact interaction, and they have to be considered in the modeling of contact problems if the substrate is micropolar and size dependent.

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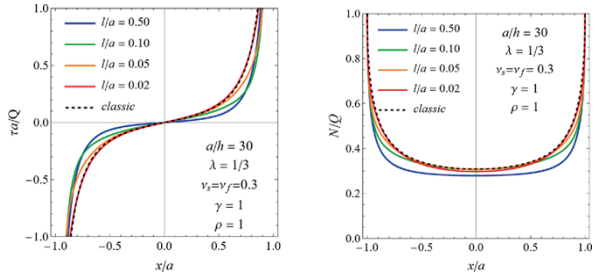
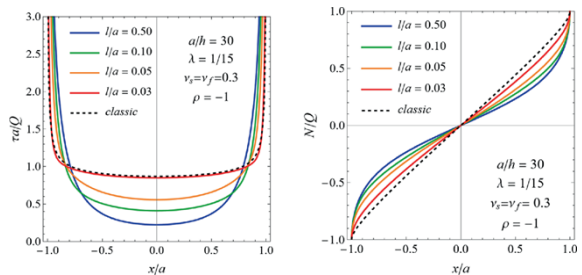
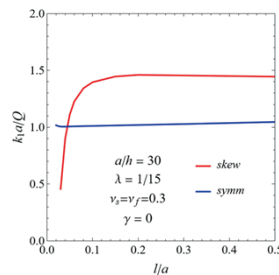
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Chapter 30

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Page no	Fig/Photo	Thumbnail	Alt-text Description
4	Fig1	<p>The diagram illustrates a mechanical system with three main components. On the left, a semicircular green shape is shown with a coordinate system (x, y) where the origin (0) is at the center of the flat base. The radius is labeled as 'a'. Two horizontal forces, Q_1 and Q_2, are applied at the ends of the flat base. Material parameters $\mu, \nu, \ell, \epsilon^0$ are indicated. In the center, a magnified square section shows stress components $\sigma_{xx}, \sigma_{yy}, \sigma_{xy}$ and moments m_{xz}, m_{yz}. On the right, a rectangular element with height h and width dx is shown, with normal forces N and $N + dN$, and shear stress τ with a moment m.</p>	<p>Diagram illustrating a mechanical system with three main components. On the left, a semicircular green shape with labeled parameters: $(\mu, \nu, \ell, \epsilon^0)$, and forces (Q_1) and (Q_2) acting horizontally. In the center, a magnified square section shows stress components $(\sigma_{xx}, \sigma_{yy}, \sigma_{xy})$ and moments (m_{xz}, m_{yz}) with arrows indicating directions. On the right, a rectangular element with height (h) and width (dx) shows forces (N) and $(N + dN)$, and shear stress (τ) with moment (m). Arrows indicate force and moment directions.</p>
6	Fig2	<p>The figure consists of two panels. The left panel plots $\tau a/Q$ against x/a for l/a values of 0.50 (blue), 0.10 (green), 0.05 (orange), and 0.02 (red). A dashed line represents the 'classic' model. The right panel plots N/Q against x/a for the same l/a values and a dashed 'classic' line. Parameters for both panels are $a/h = 30, \lambda = 1/3, \nu_s = \nu_f = 0.3, \gamma = 0, \rho = 1$.</p>	<p>Two-panel figure showing X-Y charts. Left Panel: The chart plots $(\tau a/Q)$ against (x/a) with curves for different (l/a) values: 0.50 (blue), 0.10 (green), 0.05 (orange), and 0.02 (red). A dashed line represents the "classic" model. Parameters include $(a/h = 30), (\lambda = 1/3), (\nu_s = \nu_f = 0.3), (\gamma = 0),$ and $(\rho = 1)$. Right Panel: The chart plots (N/Q) against (x/a) with similar color coding for (l/a) values and a dashed "classic" line. Parameters are $(a/h = 30), (\lambda = 1/3), (\nu_s = \nu_f = 0.3), (\gamma = 0),$ and $(\rho = 1)$.</p>

Page no	Fig/Photo	Thumbnail	Alt-text Description
6	Fig3		<p>Two X-Y charts are displayed side by side. The left chart plots $(\tau a/Q)$ against (x/a), showing curves for different values of (l/a) (0.50, 0.10, 0.05, 0.02) and a classic model. The right chart plots (N/Q) against (x/a) with similar parameters. Both charts include annotations: $(a/h = 30)$, $(\lambda = 1/\beta)$, $(\nu_s = \nu_f = 0.3)$, $(\gamma = 1)$, and $(\rho = 1)$. The curves are color-coded and labeled accordingly.</p>
7	Fig4		<p>Two-panel figure showing X-Y charts. The left panel plots $(\tau a/Q)$ against (x/a) with curves for different (l/a) values: 0.50 (blue), 0.10 (green), 0.05 (orange), and 0.03 (red), compared to a classic model (dashed black line). The right panel plots (N/Q) against (x/a) with the same color scheme and classic model comparison. Both panels include parameters: $(a/h = 30)$, $(\lambda = 1/15)$, $(\nu_s = \nu_f = 0.3)$, and $(\rho = -1)$.</p>
7	Fig5		<p>Graph showing the relationship between (l/a) on the x-axis and $(k_1 a/Q)$ on the y-axis. Two lines are plotted: a red line labeled "skew" and a blue line labeled "symm." The red line rises sharply before leveling off, while the blue line remains relatively constant. Annotations include $(a/h = 30)$, $(\lambda = 1/15)$, $(\nu_s = \nu_f = 0.3)$, and $(\gamma = 0)$.</p>